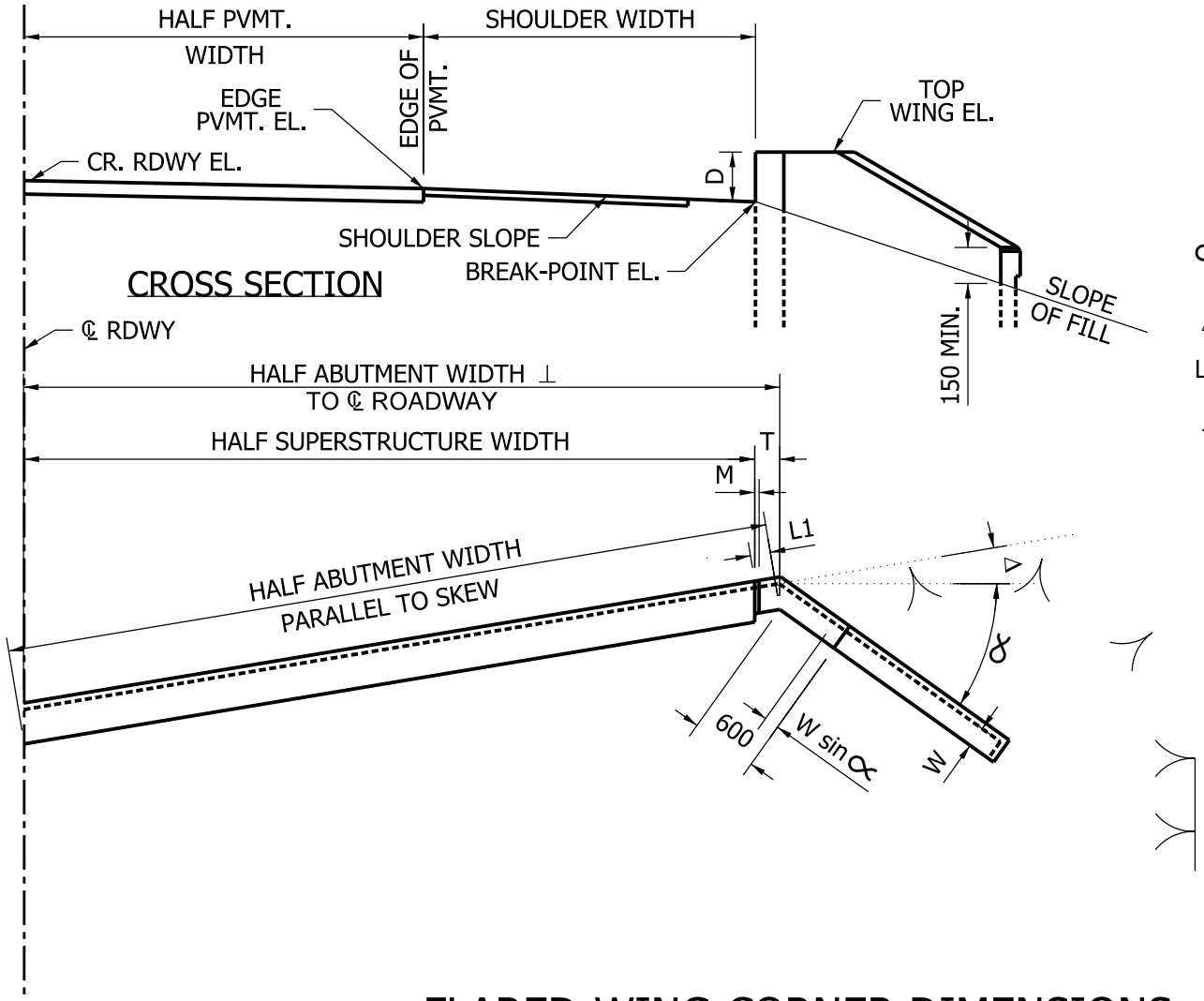

D = TOP WING EL. - BREAK-POINT EL.

∝ = ANGLE BETWEEN WING AND LINE ⊥ TO ℄ RDWY.

Δ = SKEW ANGLE

$L1 = \frac{T}{\cos \Delta}$

T & M - TO BE AS DETERMINED BY DESIGNER.

FLARED-WING-CORNER DIMENSIONS, STRUCTURE SKEWED TO LEFT

Figure 67-2C (3)